



## QUICK-RETURN MECHANISM REVISITED

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# Work In Progress: Quick-Return Mechanism Revisited

## Abstract

In this paper, the teaching and learning experiences of the author with two summer interns at one of the educational institutions in India is presented. These are the senior mechanical engineering students from two different engineering colleges in India who spent nearly two months at the institute where the author spent a 3-month sabbatical as a visiting faculty. Although these two students took the “Theory of Machines” course at their college, a complete understanding of kinematic and dynamic analyses of mechanisms such as a quick-return linkage seemed to be not realized well by them. In addition to the students from India, there are other mechanical engineering students who were taking a Design and Analysis of Mechanical Systems and Assemblies course as a directed study. The students were taught the basics of loop-closure equations pertaining to the kinematic and dynamic analysis of an example quick-return and other planar mechanisms. All these students developed an Excel based program to perform calculations and plot the various characteristics such as variation of quick return ratio as a function of the critical link lengths, kinematic and dynamic characteristics of the linkage. Studies related to partially balance the system are also under way, mostly using a CAE tool. The students modeled the linkage using the motion simulation application that is commonly available in any CAE tool such as Catia, UG-NX, NX I-DEAS, or SolidWorks. Other math tools such as MatLab Simulink, MapleSim, etc., are also available to study planar mechanism kinematics. Finally, the students in India used the available laboratory experimental apparatus to verify some of the theoretical calculations. The performance metric is a final report that included the learning outcomes and recommendations for further work.

## Introduction and literature review

The Course Learning Objectives (CLOs) of the course are:

1. *Apply* the integration of the fundamental concepts of rigid body kinematics in relative motion, solid mechanics and computer aided engineering through computational and design tools.
2. *Apply* fundamental mechanics principles to the kinematic, dynamic and fatigue stress analyses of components of planar mechanisms, subsystems and systems.
3. *Use* state-of-the-art CAE software tools to formulate, conceptualize, design, analyze, and synthesize open-ended problems pertaining to mechanical systems.
4. *Develop* strategies to improve the product and process design based on the results obtained.

In tune with the above CLOs, students the course is taught using a combination of theory using graphical and analytical methods and CAE tools such as I-DEAS or NX 7.5. However, covering

in depth theory using analytical kinematics is found to be challenging due to time constraints. On the other hand, the conventional graphical methods although limited to analysis of a mechanism in the instantaneous reference frame, students seem to realize their ease of use. In order to expand the students understanding of mechanisms, it is important to explore the same system at multiple points in time, delivering an understanding of the cycle the mechanism would go through during operation. It is common for students and researchers to explore the use of software to design and analyze mechanism cycles using CAE tools such as Catia, Unigraphics-NX, HyperWorks, NX I-DEAS or SolidWorks<sup>1-5</sup>. Also mathematical tools such as MapleSim<sup>6</sup> and Matlab<sup>7</sup> and written programs such as C<sup>++</sup>, Fortran and T-K Solver<sup>8-10</sup>, etc are also used. However, the use of these programs does not require the student to have a deeper understanding of the methods being used in the analysis. In the case of software such as C<sup>++</sup>, Matlab, and Maplesim the student does not have a visual representation of how the model behaves while in motion. Conversely, to use solid modeling and simulations software such as Solidworks and UG-NX the student is not required to have a full understanding of the methods being used in the analysis. In order to allow students to analyze an example model while still understanding the methods involved, an analysis program for a Whitworth quick-return mechanism was created in Microsoft excel. The same model was modeled in a CAE software and motion was simulated to create a reference for verification of the excel model. The model was further cross referenced to a previously published work on the use of a C<sup>++</sup> program that provided a solution to the example in question<sup>11</sup>. The example used is the Whitworth quick-return system, an uncommonly explored linkage because it has not been used in high frequency application due to fundamental difficulties with unbalance and vibrations. The Microsoft Excel program generated in this report is eventually to be expanded upon with the interest of exploring methods to balance the system, reducing vibrations in and expanding on the applications of this mechanism. A physical model was subsequently used by the students in India to acquire data and verify the authenticity of the equations and models.

## Analysis

The model analyzed in this paper was replicated from a previous paper that explored the use of C<sup>++</sup> programming to fully model a Whitworth Quick return mechanism published by Matt Campbell and Stephan Nestinger<sup>11</sup>. In the interests of cross referential verification, the model simulated in this paper is the same as the model in the previously published paper.

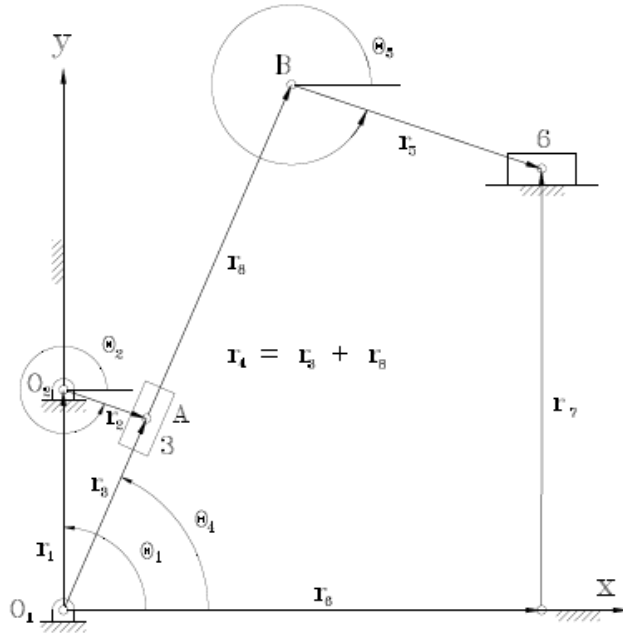


Figure 1. Vector representation of the Whitworth Quick Return Mechanism <sup>11</sup>

The equations for Kinematic analysis that are presented below were originally presented by Campbell and Nestinger<sup>11</sup>.

**Position Analysis:**

The displacement analysis can be formulated by following equations:

$$r_1 + r_2 = r_3 \tag{1a}$$

$$r_3 + r_8 + r_5 = r_6 + r_7 \tag{1b}$$

Using complex numbers, Equations 1 and 2 become

$$r_3 r_1 e^{i\theta_1} + r_2 e^{i\theta_2} = r_3 e^{i\theta_3} \tag{2a}$$

$$e^{i\theta_3} + r_8 e^{i\theta_8} + r_5 e^{i\theta_5} = r_6 e^{i\theta_6} + r_7 e^{i\theta_7} \tag{2b}$$

Here, link lengths  $r_1, r_2, r_4, r_5, r_7$  and angular positions  $\theta_1, \theta_6, \theta_7$  are constants.

Now, equ.1 becomes

$$r_3 e^{i\theta_4} = r_1 e^{i\theta_1} + r_2 e^{i\theta_2} \tag{3a}$$

And since,  $r_4 = r_3 + r_8$ .

Therefore, equ.4 can be written as,

$$r_4 e^{i\theta_4} + r_5 e^{i\theta_5} - r_6 e^{i\theta_6} = r_7 e^{i\theta_7} \tag{3b}$$

Using Euler's equation,

$$e^{i\theta} = \cos \theta + i \sin \theta,$$

$$r_3 \cos \theta_4 = r_1 \cos \theta_1 + r_2 \cos \theta_2 \tag{4a}$$

$$r_3 \sin \theta_4 = r_1 \sin \theta_1 + r_2 \sin \theta_2 \tag{4b}$$

Squaring Equations 4, adding them together, we get

$$r_3 = \sqrt{(r_1 \cos \theta_1 + r_2 \cos \theta_2)^2 + (r_1 \sin \theta_1 + r_2 \sin \theta_2)^2} \tag{5}$$

Dividing Equation 4b by Equation 4a and simplifying gives

$$\theta_4 = \tan^{-1} \left( \frac{r_1 \sin \theta_1 + r_2 \sin \theta_2}{r_1 \cos \theta_1 + r_2 \cos \theta_2} \right) \quad (6)$$

Now, equ. (3b) can be written as

$$r_6 e^{i\theta_6} - r_5 e^{i\theta_5} = r_4 e^{i\theta_4} - r_7 e^{i\theta_7} \quad (7)$$

Since, the right hand side of equ. 7 is constant, let us consider

$$r e^{i\theta} = r_4 e^{i\theta_4} - r_7 e^{i\theta_7}$$

This is used in further calculations. Now, equ.7 becomes

$$r_6 \cos \theta_6 - r_5 \cos \theta_5 = r \cos \theta \quad (8a)$$

$$r_6 \sin \theta_6 - r_5 \sin \theta_5 = r \sin \theta \quad (8b)$$

Solving Equations 8 for  $r_6$  gives

$$r_{6a} = \frac{r \cos \theta + r_5 \cos \theta_5}{\cos \theta_6} \quad (9a)$$

$$r_{6b} = \frac{r \sin \theta + r_5 \sin \theta_5}{\sin \theta_6} \quad (9b)$$

Where equ. (9a) is used when  $\cos \theta_6 > 0$  and equ. 9b is used when  $\cos \theta_6 = 0$ . Substituting Equ. (9a) into Equ. (8b) gives

$$\sin(\theta_5 - \theta_6) = \frac{r \cos \theta \sin \theta_6 - r \sin \theta \cos \theta_6}{r_5} \quad (10)$$

Solving for  $\theta_5$ , we find

$$\theta_{5a} = \theta_6 + \sin^{-1} \left( \frac{r \cos \theta \sin \theta_6 - r \sin \theta \cos \theta_6}{r_5} \right) \quad (11a)$$

$$\theta_{5b} = \theta_6 + \pi - \sin^{-1} \left( \frac{r \sin \theta \cos \theta_6 - r \cos \theta \sin \theta_6}{r_5} \right) \quad (11b)$$

Knowing all of the angular positions and the length of  $r_6$ , we can find the position of the output slider, link6, using

$$P_6 = r_4 + r_5 \quad (12)$$

### Velocity Analysis:

The Velocity analysis can be formulated by taking the time derivative of equ. (2) which is as follows:

$$\dot{r}_3 e^{i\theta_4} + r_3 i \omega_4 e^{i\theta_4} = \dot{r}_1 e^{i\theta_1} + r_1 i \omega_1 e^{i\theta_1} + \dot{r}_2 e^{i\theta_2} + r_2 i \omega_2 e^{i\theta_2} \quad (13a)$$

$$\dot{r}_6 e^{i\theta_6} + r_6 i \omega_6 e^{i\theta_6} = \dot{r}_4 e^{i\theta_4} + r_4 i \omega_4 e^{i\theta_4} + \dot{r}_5 e^{i\theta_5} + r_5 i \omega_5 e^{i\theta_5} - \dot{r}_7 e^{i\theta_7} - r_7 i \omega_7 e^{i\theta_7} \quad (13b)$$

Here,  $\dot{r}_1 = \dot{r}_2 = \dot{r}_4 = \dot{r}_5 = 0$  (as these links are assumed to be rigid members),  $\omega_1 = 0$  (since link 1 is rigid),  $\omega_6 = \omega_7 = 0$  and  $\theta_6 = 0$  as links 6 and 7 are assumed to be non-rotating imaginary members, and  $\dot{r}_7 = 0$  because the output slider 6 is assumed to remain on the ground at all times. Taking these considerations, we have

$$\dot{r}_3 e^{i\theta_4} + r_3 i \omega_4 e^{i\theta_4} = r_2 i \omega_2 e^{i\theta_2} \quad (14a)$$

$$\dot{r}_6 = r_4 i \omega_4 e^{i\theta_4} + r_5 i \omega_5 e^{i\theta_5} \quad (14b)$$

Solving for  $\omega_4$ , we have

$$\omega_4 = \frac{r_2 \omega_2 \cos \theta_2 \cos \theta_4 + r_2 \omega_2 \sin \theta_2 \sin \theta_4}{r_3} \quad (15)$$

Substituting equ. (15) into either the real or imaginary equ. of equ. (14a) Equ.s (14) can be written as,

$$\dot{r}_6 = -r_4 \omega_4 \sin \theta_4 - r_5 \omega_5 \sin \theta_5 \quad (16a)$$

$$0 = r_4 \omega_4 \cos \theta_4 + r_5 \omega_5 \cos \theta_5 \quad (16b)$$

From equ. (16b),

$$\omega_5 = -\frac{r_4 \omega_4 \cos \theta_4}{r_5 \cos \theta_5} \quad (17)$$

Thus, velocity of output slider can be found out by using

$$V_6 = \dot{r}_6 + \dot{r}_7 \quad (18)$$

Since, the vertical component of velocity,  $V_{6y} = 0$ , so  $V_{6x} = \dot{r}_6$

### Acceleration Analysis:

The acceleration analysis has been formulated by taking the first time derivative of equ.s (14) Breaking up equ. (14a) into its real and imaginary parts, we have

$$\dot{r}_3 = \sqrt{(r_2 \omega_2 \cos \theta_2 - r_3 \omega_4 \cos \theta_4)^2 + (r_3 \omega_4 \sin \theta_4 - r_2 \omega_2 \sin \theta_2)^2} \quad (19)$$

The angular acceleration of link 4 was found out as

$$\alpha_4 = \frac{r_2}{r_3} \{-\omega_2^2 \cos(\theta_2 - \theta_4) + \alpha_2 \sin(\theta_2 - \theta_4)\} - 2 \frac{\dot{r}_3}{r_3} \omega_4 \quad (20)$$

The linear acceleration of output slider is given as

$$\ddot{r}_6 = -r_4\alpha_4 \sin \theta_4 - r_4\omega_4^2 \cos \theta_4 - r_5\alpha_5 \sin \theta_5 - r_5\omega_5^2 \cos \theta_5 \quad (21)$$

$$0 = r_4\alpha_4 \cos \theta_4 - r_4\omega_4^2 \sin \theta_4 + r_5\alpha_5 \cos \theta_5 - r_5\omega_5^2 \sin \theta_5 \quad (22)$$

From equ. (22), we get,

$$\alpha_5 = \frac{r_4(\omega_4^2 \sin \theta_4 - \alpha_4 \cos \theta_4) + r_5\omega_5^2 \sin \theta_5}{r_5 \cos \theta_5} \quad (23)$$

Substituting  $\alpha_5$  in equ. (21), we get acceleration of the output slider i.e.

$$a_6 = \ddot{r}_6 + \ddot{r}_7 \quad (24)$$

Since the vertical component of acceleration,  $a_{6y} = 0$  so,  $a_{6x} = \ddot{r}_6$

Dynamic force analysis equations can also be programmed in Excel following the same process. This, together with balancing is in progress.

Following sections present the results of the analytical study and discuss the learning outcomes of the students.

## Results and Discussion

### Analytical model

The first analysis completed in the Excel programs was the variance of the quick return ratio, or QRR, of the inverted slider crank loop as a function of its component links which were individually varied. The result visually demonstrated to the students what had been shown in the equations. As shown in figure 2 below, for each case the system has an asymptotic relationship to its bounding conditions, i.e., the link lengths can never be equal or else the QRR approaches infinity as the link lengths approach each other.

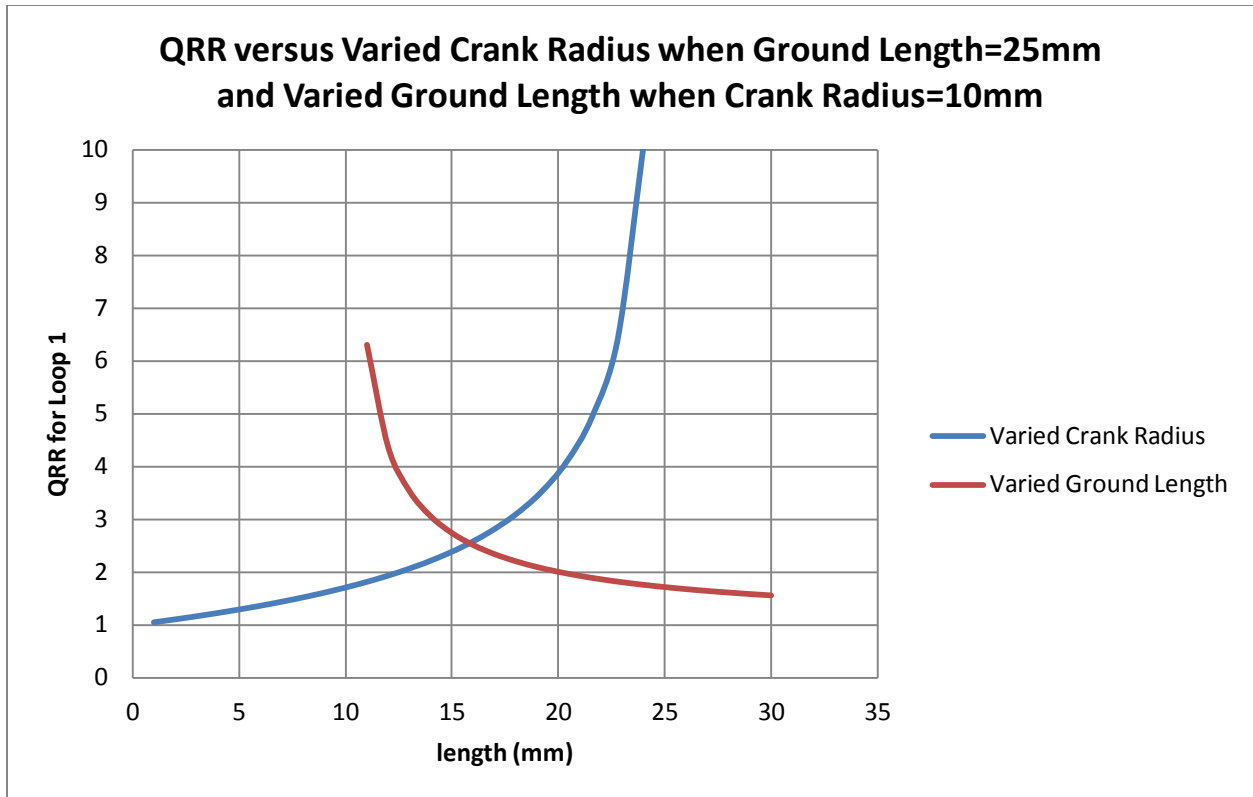


Figure 2. QRR as ground link and crank link lengths are varied

The result of varying the QRR by changing the link lengths was further explored by examining the components in the first loop. In figures 3 and 4 below the baseline is the behavior of the model that was replicated from Campbell and Nestinger<sup>11</sup>, and by varying the link lengths alters the behavior of the respective links within the first loop. Figure 3 examines the changes in the slider position relative the ground pivot point. In all cases the slider oscillates between the difference and the sum of the crank and ground link lengths. Varying the ground link will vary the average value, while varying the crank length will vary the amplitude of the oscillation. Figure 4 examines the effects of QRR on the angle of the output arm link 4. As the arm continues to oscillate about the vertical position, the average value for all cases remains 90 degrees, or  $\frac{\pi}{2}$  radians for all cases. Varying the QRR has the same effect at the extremes for variance of either link length. As the length difference approaches zero, the slope of theta 4 approaches infinity at the bottom dead center position. Conversely, as either the ground length (L1) approaches infinity or the crank length (L2) approaches zero, the amplitude of the oscillation approaches zero.



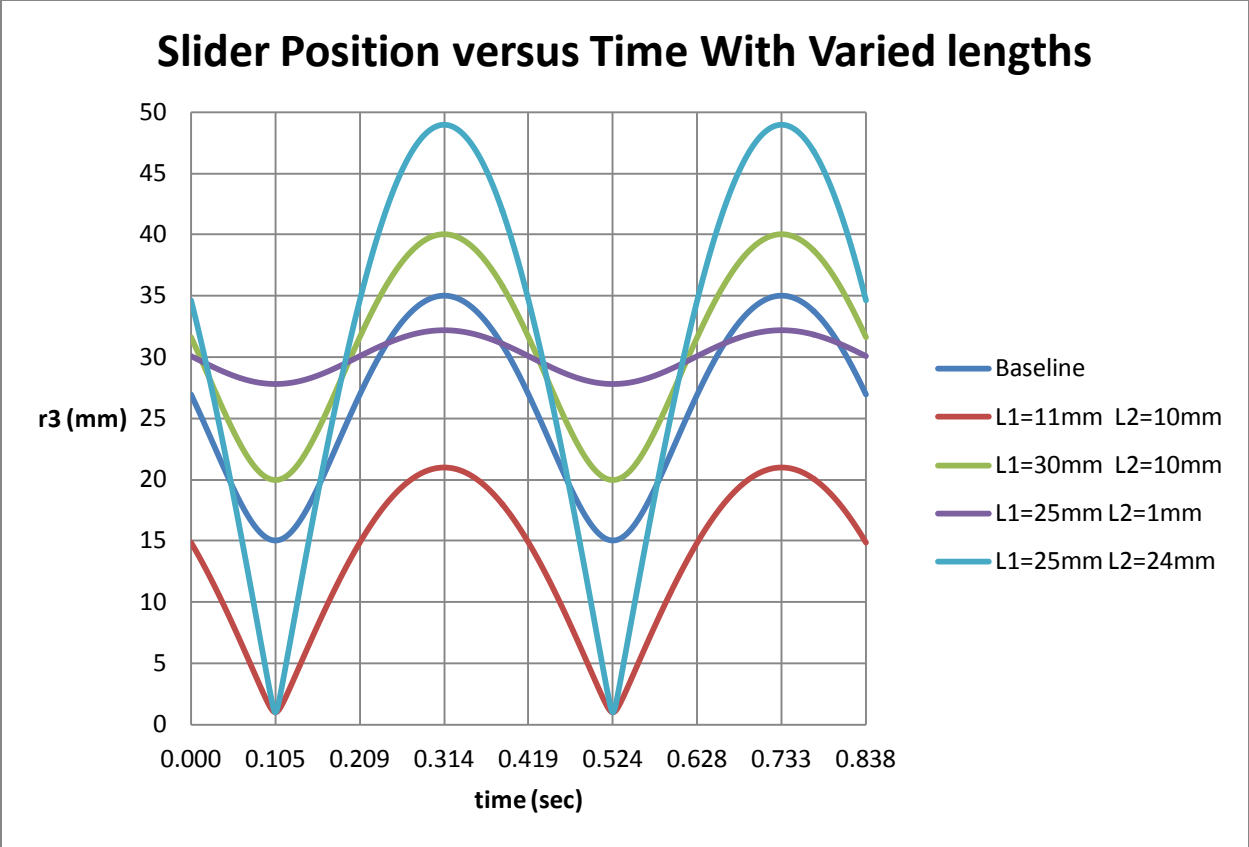


Figure 3. Slider Position Relative to Ground Pivot as QRR approaches extremes

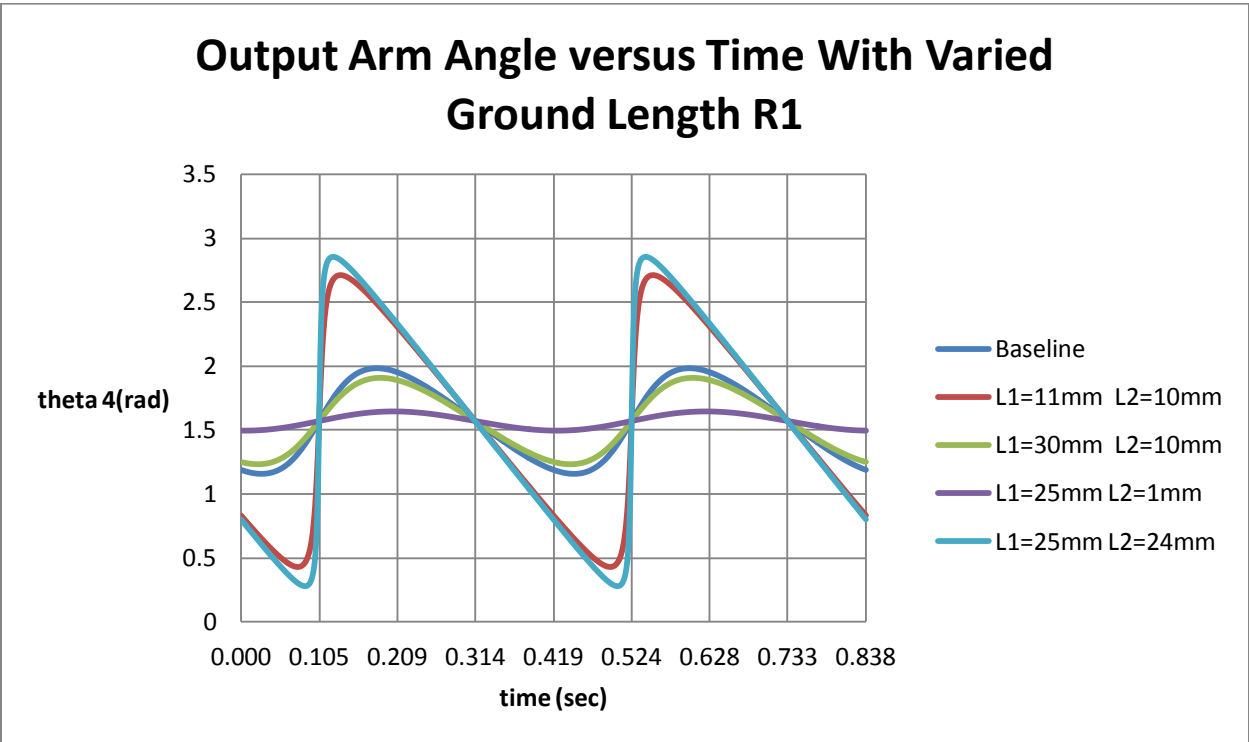


Figure 4. Loop 1 Output Arm Angle as QRR approaches extremes

The first point of comparison between the excel program, the simulation results from the CAE simulation, in this case Unigraphics NX7.5, and the results published by Campbell and Nestinger<sup>11</sup> was the velocity analysis of the output slider link 6. As shown in Figures 5 through 7, the results across all three are the same.

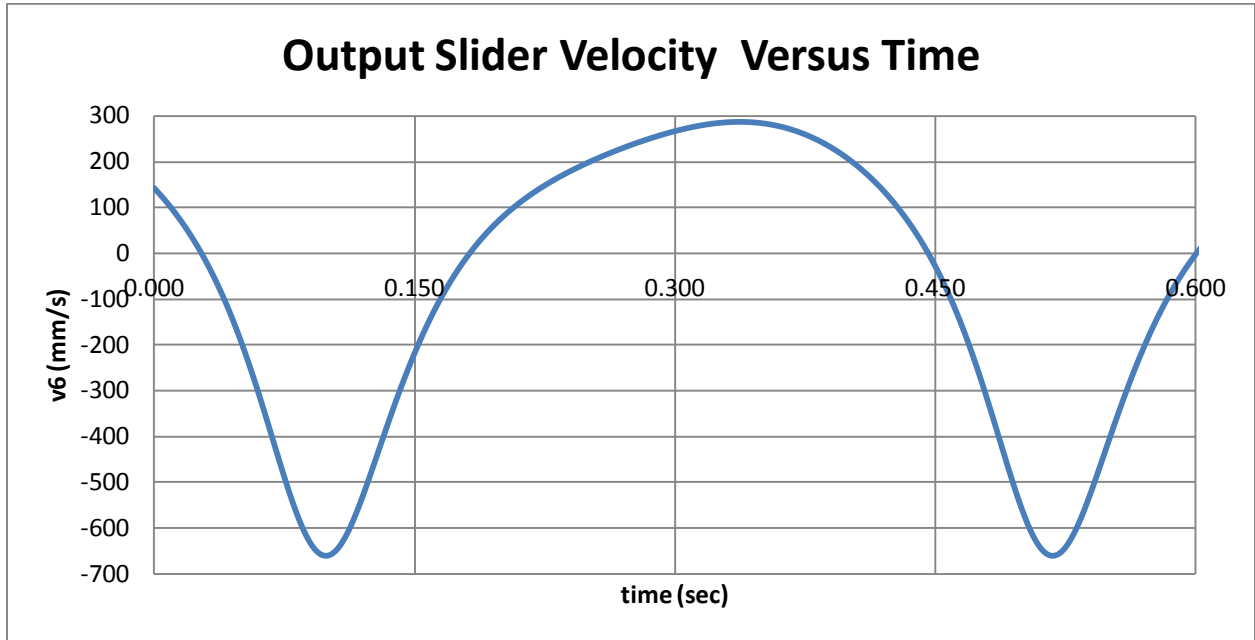


Figure 5. Excel Result for Output Slider Velocity versus Time

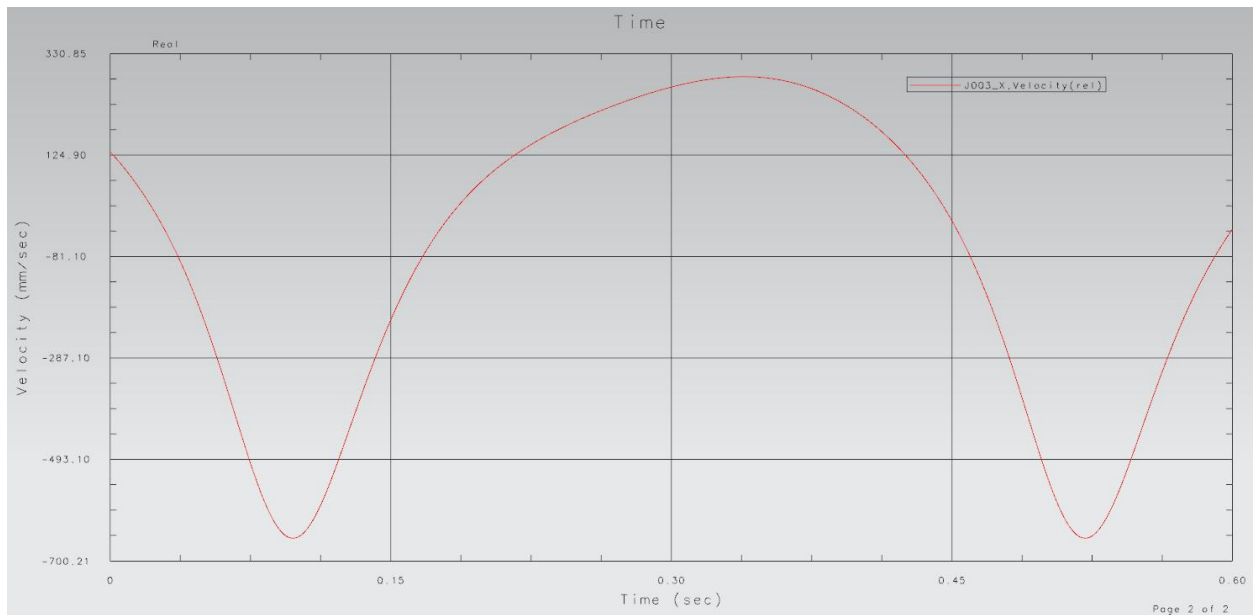


Figure 6. Unigraphics NX7.5 Result for Output Slider Velocity versus Time

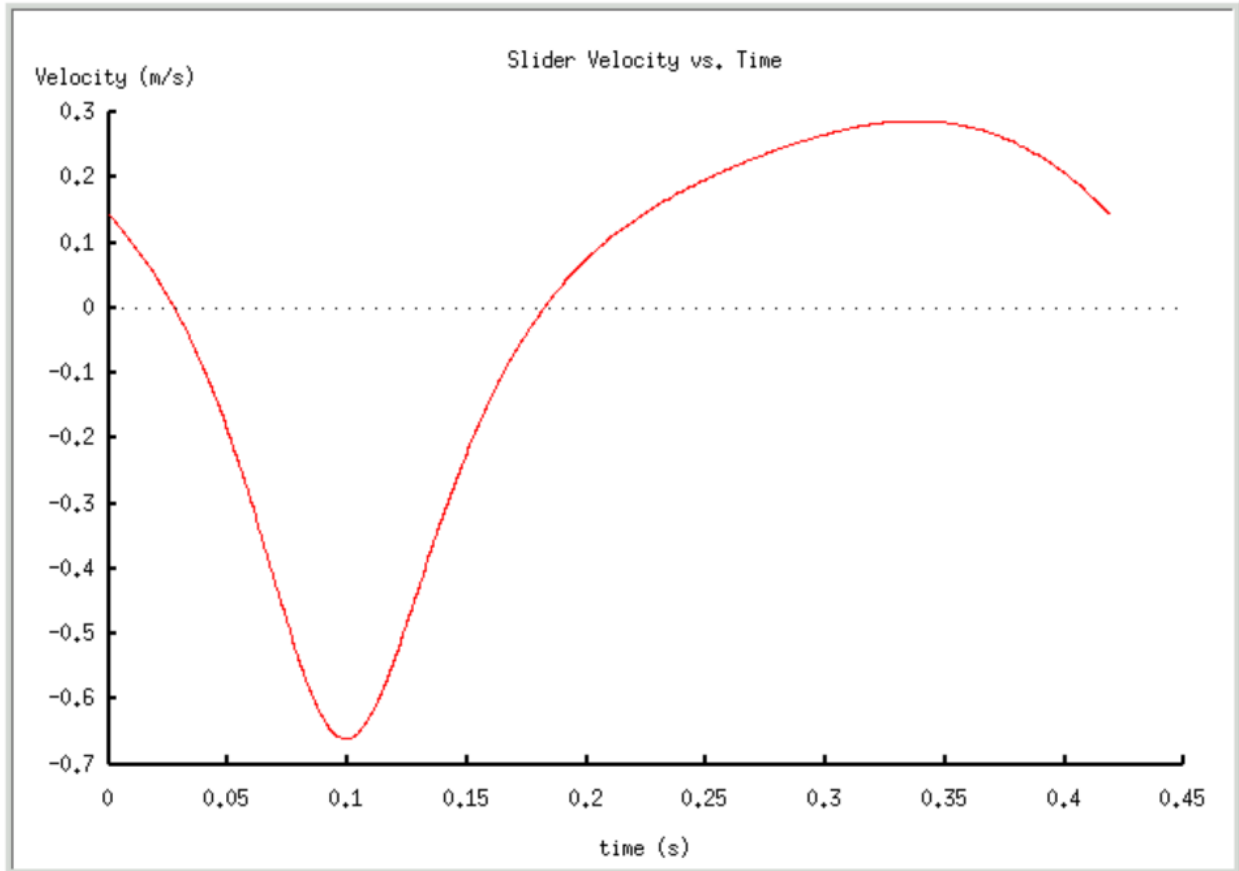


Figure 7. Output Slider Velocity versus Time as published by Campbell and Nestinger<sup>11</sup>

In the course of verification between Campbell findings, the output from the Unigraphics NX7.5, and the result of the excel program the students noticed that error and discontinuity were introduced close to the so called “dead center positions” if simply following the equations presented by Campbell and Nestinger<sup>11</sup>. As these discontinuities were corrected the students gained an improved understanding of the time variance in the system. Similar processes were completed in calculating the accelerations of each link, and subsequently the forces on each link and at each joint. This will be presented in the final draft.

### Physical Model and Data Acquisition System:

In order to better understand the kinematics of the mechanism, experiments were done at the Indian Institute of Technology – Gandhinagar to analyze the behavior of the system and the various conditions associated with it. The tests done in the laboratory on a physical model with data acquisition and sensors helped the students in better visualization of the actual system and the associated data acquisition and measurements. This also helped the students to better understand the various types of sensors used for the measurement. A bread-board model at Kettering University is also available that can be modified and assembled to demonstrate the motion of the quick-return and several other planar mechanisms. The quick-return model used at the Indian institution is shown in figure 8 below.

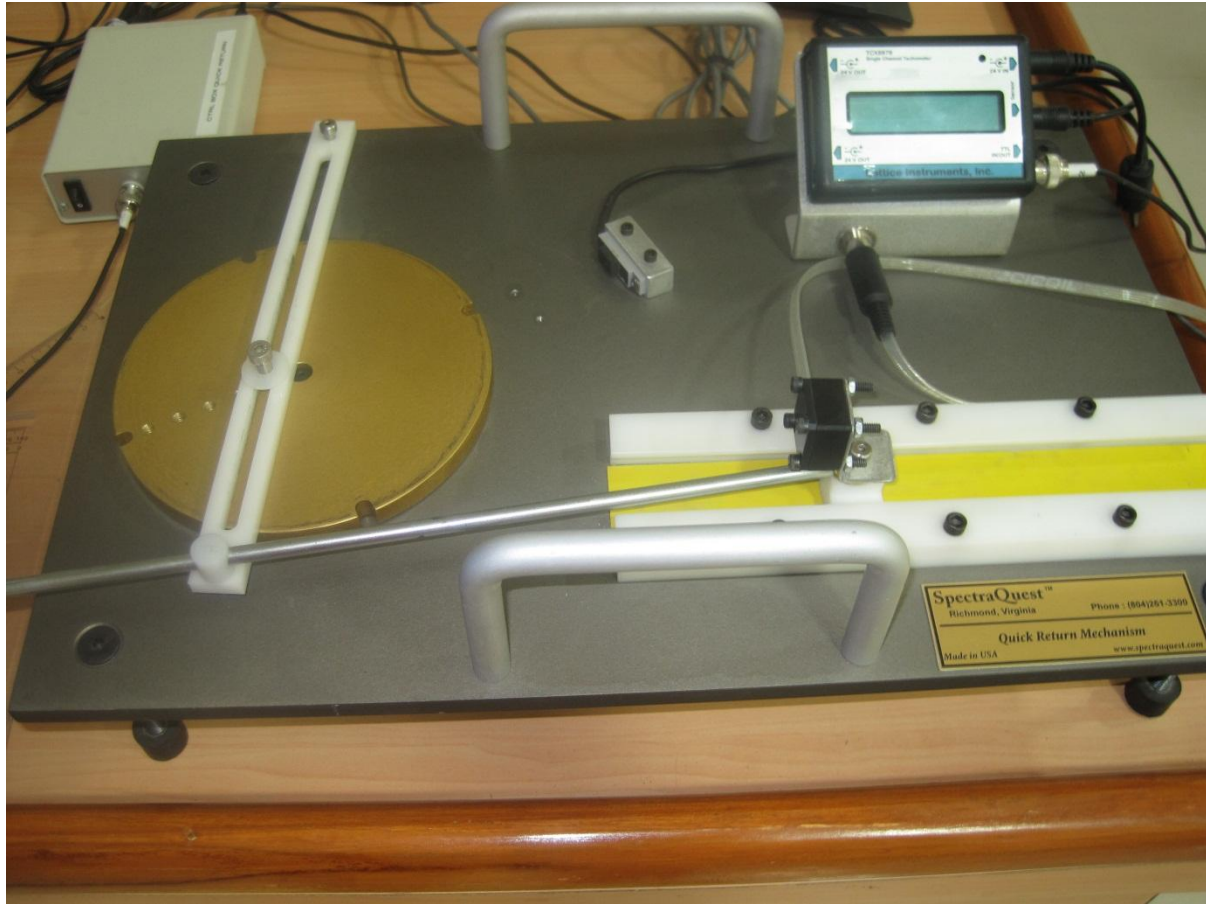


Figure 8. Physical model of a Whitworth Quick Return system Used

The major components of this apparatus are the motor, Quick return links (crank, connecting rod, slider, fixed link), sensors (Accelerometer, Tachometer), and the data acquisition system

#### **Motor:**

The motor is a constant speed type that derives power from the electrical source. The motor has a provision to be driven at different constant speeds. The motor is coupled with the rotating disc.

#### **Quick Return Links:**

The various links are assembled and are driven by the motor at various speeds. In this apparatus the crank radius can change for 5 different position of crank. For this various crank radius the quick return ratio is calculated.

#### **Accelerometer Sensors:**

The accelerometer sensor that is used in this apparatus is capacitive type. These capacitors operate in a bridge circuit, along with two fixed capacitors, and alter the peak voltage generated by an oscillator when the structure undergoes acceleration. Detection circuits capture the peak voltage, which is then fed to a summing amplifier that processes the final output signal.

**Tachometer:**

The tachometer is a device used to measure the speed of a rotating object. The tachometer used here is of inductive type. The variation of air gap induces a pulse which is counted by a counter and the rpm is counted.

**Data Acquisition System:**

The data acquisition system takes the analog output from the various sensors and converts them into digital values by means of Analog to Digital converter. This digital analog value is fed into the processing unit which does the required process and gives the output to the display unit. The computer uses software called KDM (Kinematics and Dynamics Of Machines). The needed values and their characteristic curves are plotted by the software and the output is recorded.

The charts as shown below in figure 9 were the outputs when crank length was 12.7 mm and the system is being driven at the crank with a rotating speed of 70 rpm.

KDM –Software: - The DATA ACQUISITION system takes the analog input from the sensors and converts it to digital signals and processes it further using the KDM software to generate the various characteristic curves. The parameters that are plotted have not given any units since they just the voltage equivalent of the output. The maximum and minimum values of the displacement, velocity and acceleration can be observed from the graph. These plots do not show results for full cycle of operation.

Various experiments have been performed by changing the input speed and the crank radius to plot the linear velocity and acceleration of the slider which changes with respect to variation in crank speed and crank radius. The students also learned that the graphs although look smooth have some noise from signals and also from vibrations caused due to moving links that are not in real life rigid.

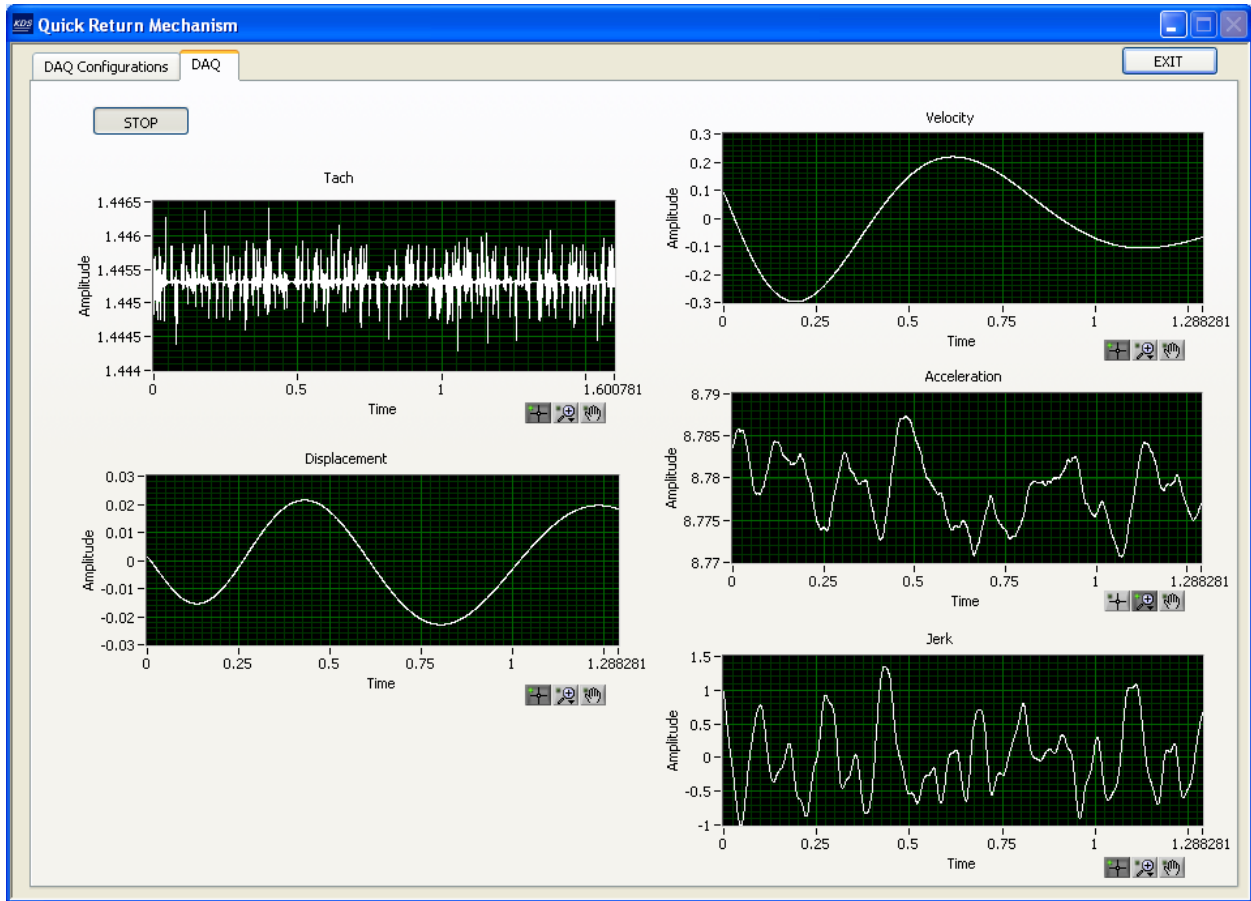


Figure 9. Outputs of physical data acquisition lab setup

As mentioned above, the physical model allowed students to vary the crank length and measure the resultant change on the output slider while observing the changing behavior of the system. In the absence of a physical model, students were able to conduct similar experimentation for the inverted slider crank loop of the complete mechanism by using the Catia CAE tool. This model will be presented in the final manuscript of the paper. Another example of such a CAE model using UG-NX 7.5 is shown in Figure 10 below. The dimensions used for this model are the same as those presented in the literature.

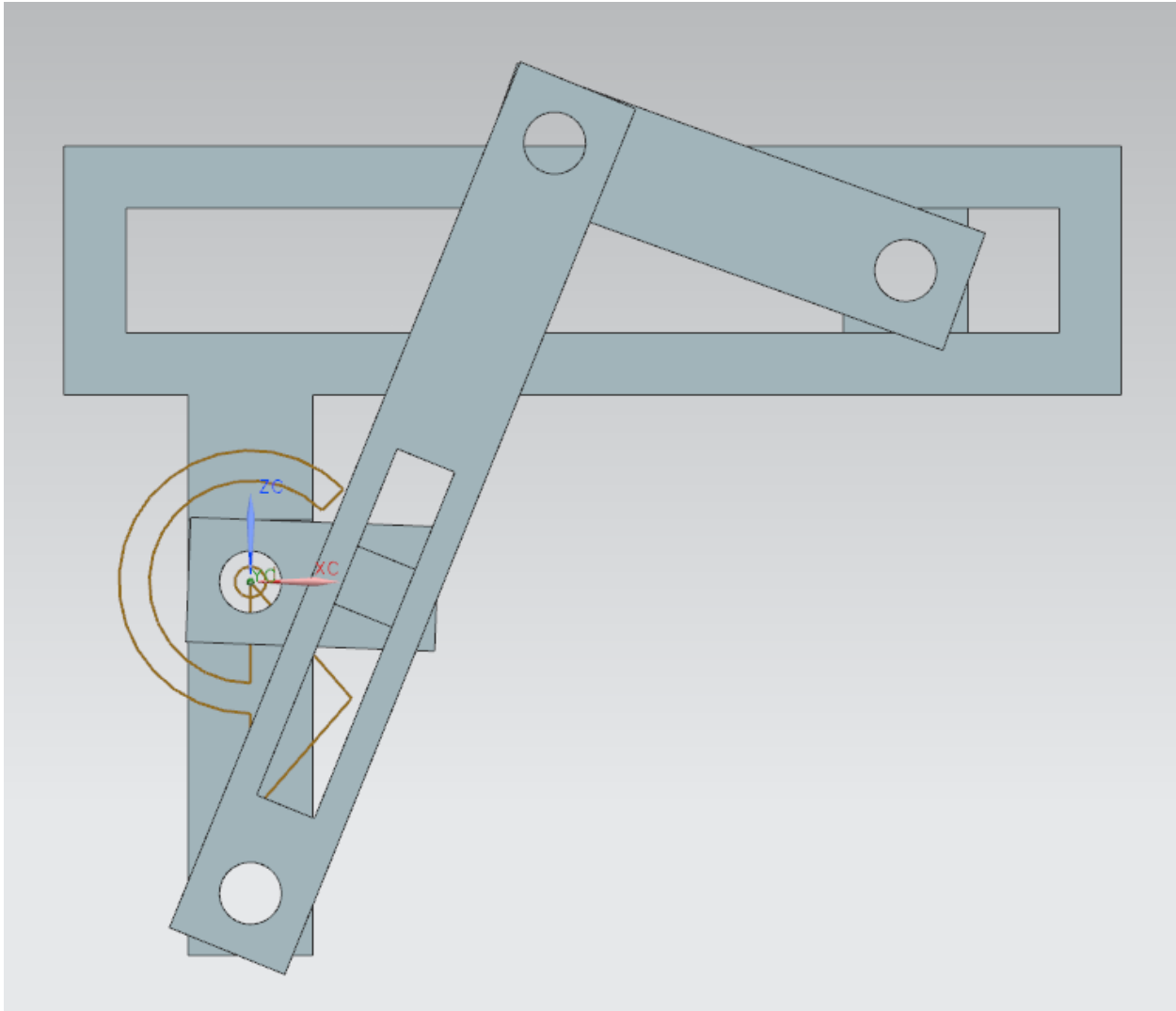


Figure 10. Whitworth Quick Return modeled in Unigraphics NX7.5 for simulation

These CAE models allow the student to construct similar experiments and generate several plots while retaining the visual demonstration of the apparatus.

### **Conclusion:**

Based on the example work presented in this paper, the students have demonstrated an increased depth of understanding of planar mechanism theory via the creation and verification of their graphical and analytical models using math and CAE tools such as Excel program and NX while also retaining a solid grasp of the physical system via either the data acquisition apparatus or the virtual CAE model. In doing so, they have explored and defined the various limiting link conditions (dead center position) of the Whitworth Quick Return system and the ramifications of the said conditions. The variance of the ground link length and/or the crank length and thus the QRR alters the system behavior and increases the understanding of the applicability of the linkage for applications involving a quick return cycle. Further studies to partially balance the linkage will be undertaken by the future students of this class. The learning outcomes written by

the students indicate that they learned the theory well when complimented with use of a simulation tool such as a CAE or a MATH tool. Further, they appreciated the use of a real experimental apparatus, which enabled them to understand better the measurement system and their limitations based on a comparison of the theoretical and experimental results.

Although in this paper only a quick return mechanism is presented, other planar mechanisms using higher pairs (cams and gears) are also studied using both graphical and analytical methods, as well as, analysis using a simulation tool such as UG-NX. Integration of all the learning tools enable the students to learn better and just in time. The assessment tools used were the homework, laboratory reports and a comprehensive examination covering all aspects of planar mechanisms.

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